

# Space Nuclear Power Sources

(focus on radioisotope devices)

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Space NPS Seminar, EC JRC ITU  
Karlsruhe, 15 July 2004

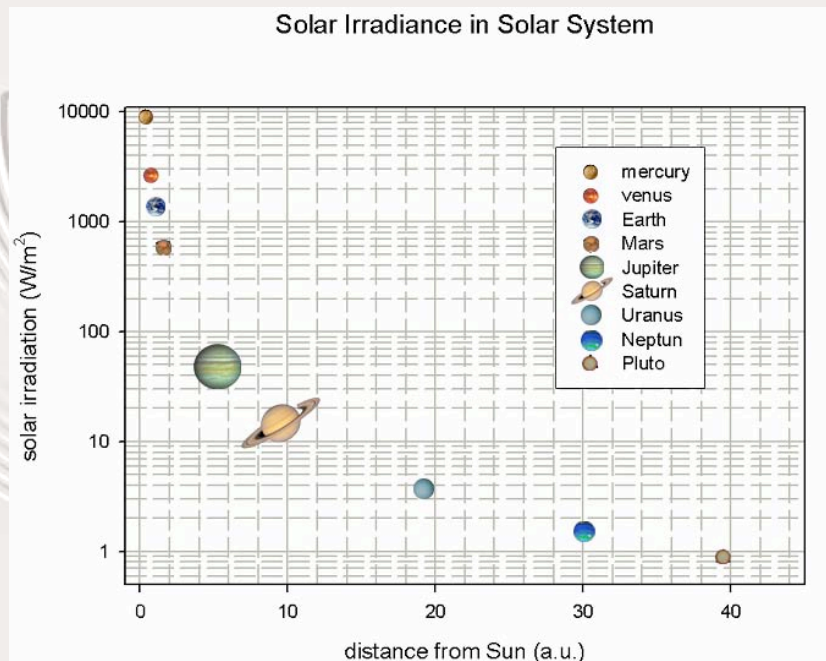


## Outline

- ④ **What are space NPS?**
- ④ **International Situation:**
  - Status quo and Plans*
  - ④ US Situation & NPS devices
  - ④ RU Situation & NPS devices
  - ④ Others
- ④ **European Situation**
  - ④ At national level
  - ④ Defence sector
  - ④ ESA
- ④ **NPS and Planned European Exploration Missions**
- ④ **Potential role of EC JRC**



# 1.1 Why Nuclear Power in space?



## 1.2 Subject of Discussion

- ⌚ **“Space nuclear power source” is the overarching term for**
  - ⌚ space nuclear reactors (1 available flight proven system (R), 5-6 kWe)
  - ⌚ radioisotope sources (heating units, thermoelectric generators)
    - ⌚ flight proven US system (GPHS RTG, 44 launched on 25 missions, 300 We)
    - ⌚ small Russian system (some We)
- ⌚ **Space reactors are small fission reactors, using the fission heat to generate electricity (nuclear electric systems) or heat up directly propellant (nuclear thermal propulsion).**
- ⌚ **Radioisotope systems use the natural decay heat of radioactive isotopes as such or to converted it into electricity.**
- ⌚ **“Nuclear Propulsion” is the overarching term for**
  - ⌚ Nuclear electric propulsion (NEP) systems  
(nuclear generated heat-electricity-electric thrusters)
  - ⌚ Nuclear thermal propulsion (NTP) systems  
(nuclear generated heat-direct propellant acceleration)



## 2.1 International Situation: US

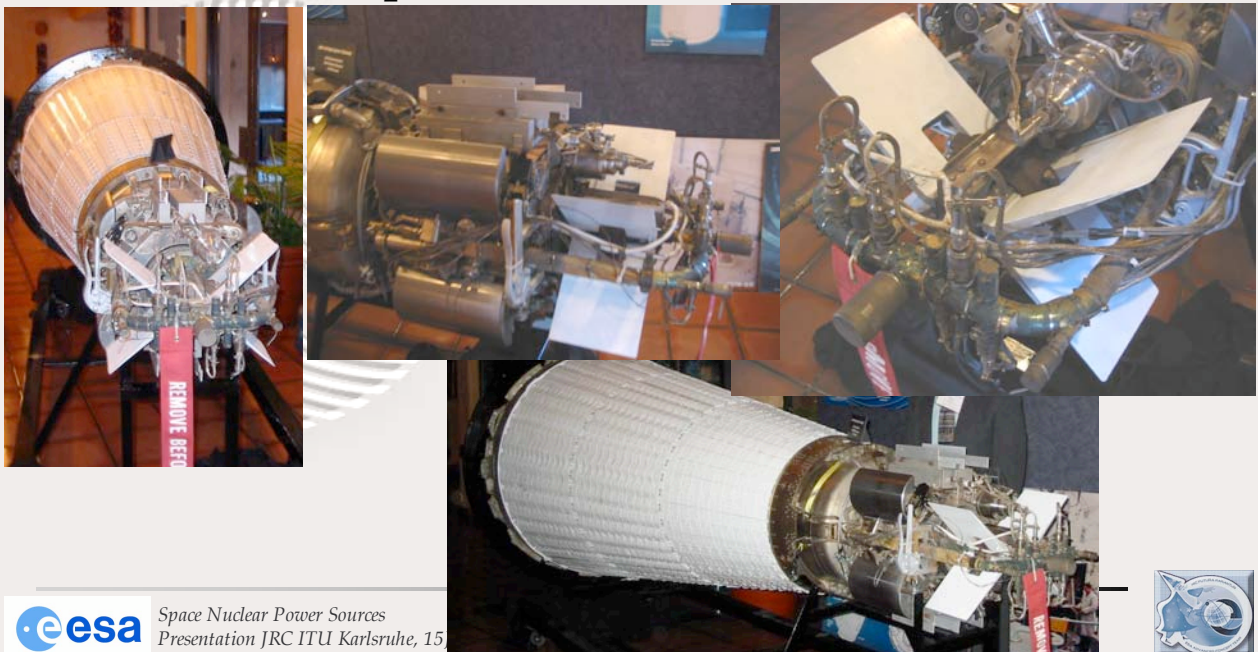
### ④ Current Status – US situation

- ④ Early decision in favour of RTGs
  - first system flown in 1961, 44 RTGs flown on 25 missions;
- ④ One flown small reactor system (1965, operated for 43 days);
- ④ Cyclic interest led to considerable investments but after 1965 to no further reactor flight model
  - ~0.5 B\$<sub>1990</sub> for SP-100 reactor;
  - inventory of subcomponent developments available;
- ④ Large nuclear thermal propulsion project (NERVA); 1960s (end: 1972/73) large total investment (initially mainly defence related, Mars mission)
- ④ Closure of Pu-238 production facility (Savannah River Site) in 1980s
  - NASA missions rely on Russian Pu-238 (agreements 1992-2002, 2004-2009)
  - US inventory reserved for national security applications



## 2.2. International Situation: US

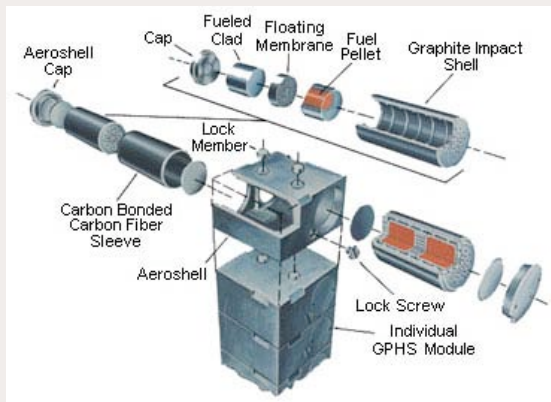
### ④ Current and past US NPS devices: 1965 reactor





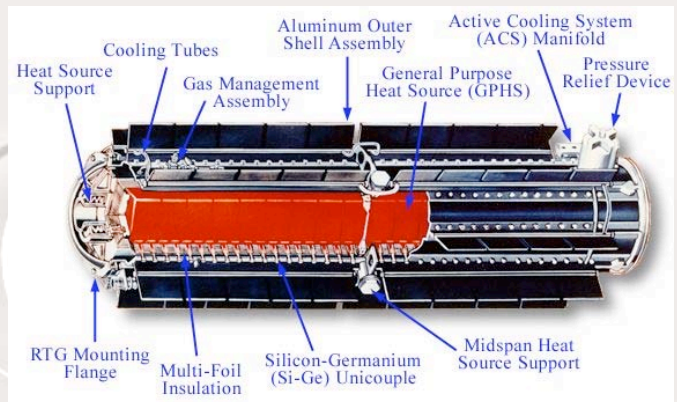
## 2.3 International Situation: US

### ⊕ Current and past US NPS devices: RTG, RHU



General Purpose Heat Source (GPMS)

**GPMS:** thermal power @BoL: ~220-240 W  
 Mass: 1.45 kg  
 Dimensions: 9.32x9.72x5.31 cm  
 PuO<sub>2</sub> mass: ~0.6 kg



General Purpose Heat Source integrated into Radioisotope Thermal Generator (RTG)

**RTG:** Thermal power BoL: ~4264 W  
 Electric Power BoL: ~285 W  
 Mass: 56 kg  
 Dimensions: 114 cm / 42 cm  
 PuO<sub>2</sub> mass: ~10.8 kg

Images: NASA, DoE



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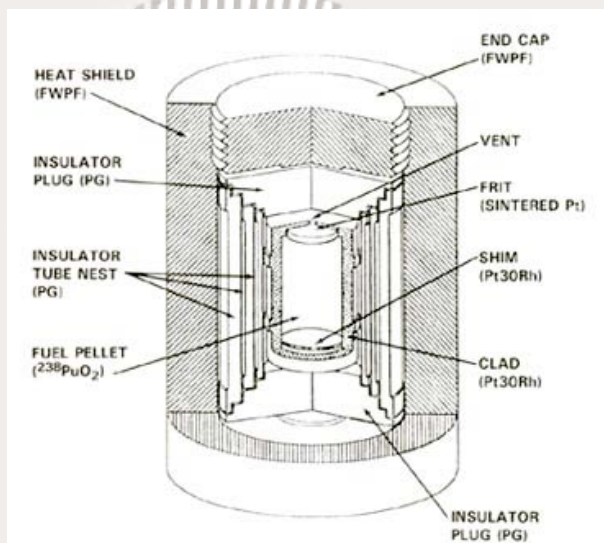
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## 2.4 International Situation: US

### ⊕ Current and past NPS devices: RHU



**RHU:** ~2.7 g PuO<sub>2</sub>  
 Platinum-30 Rhodium alloy

RTG - fuel composition

| Fuel component    | weight % at launch | $T_{1/2}$ (years) | Bq/g                 | Bq/RTG               |
|-------------------|--------------------|-------------------|----------------------|----------------------|
| <sup>236</sup> Pu | 0.000001           | 2.851             | $2.0 \times 10^{13}$ | $2.2 \times 10^{09}$ |
| <sup>238</sup> Pu | 70.810000          | 87.750            | $3.6 \times 10^{11}$ | $4.8 \times 10^{15}$ |
| <sup>239</sup> Pu | 12.859000          | 24131.000         | $2.3 \times 10^{09}$ | $3.2 \times 10^{12}$ |
| <sup>240</sup> Pu | 1.787000           | 6569.000          | $8.4 \times 10^{09}$ | $1.6 \times 10^{12}$ |
| <sup>241</sup> Pu | 0.168000           | 14.400            | $3.8 \times 10^{12}$ | $6.9 \times 10^{13}$ |
| <sup>242</sup> Pu | 0.111000           | 375800.000        | $1.5 \times 10^{08}$ | $1.8 \times 10^{09}$ |
| other             | 2.413000           |                   |                      |                      |
| Oxygen            | 11.852000          | NA                | NA                   |                      |
| total             | 100                | NA                | NA                   | $4.9 \times 10^{15}$ |



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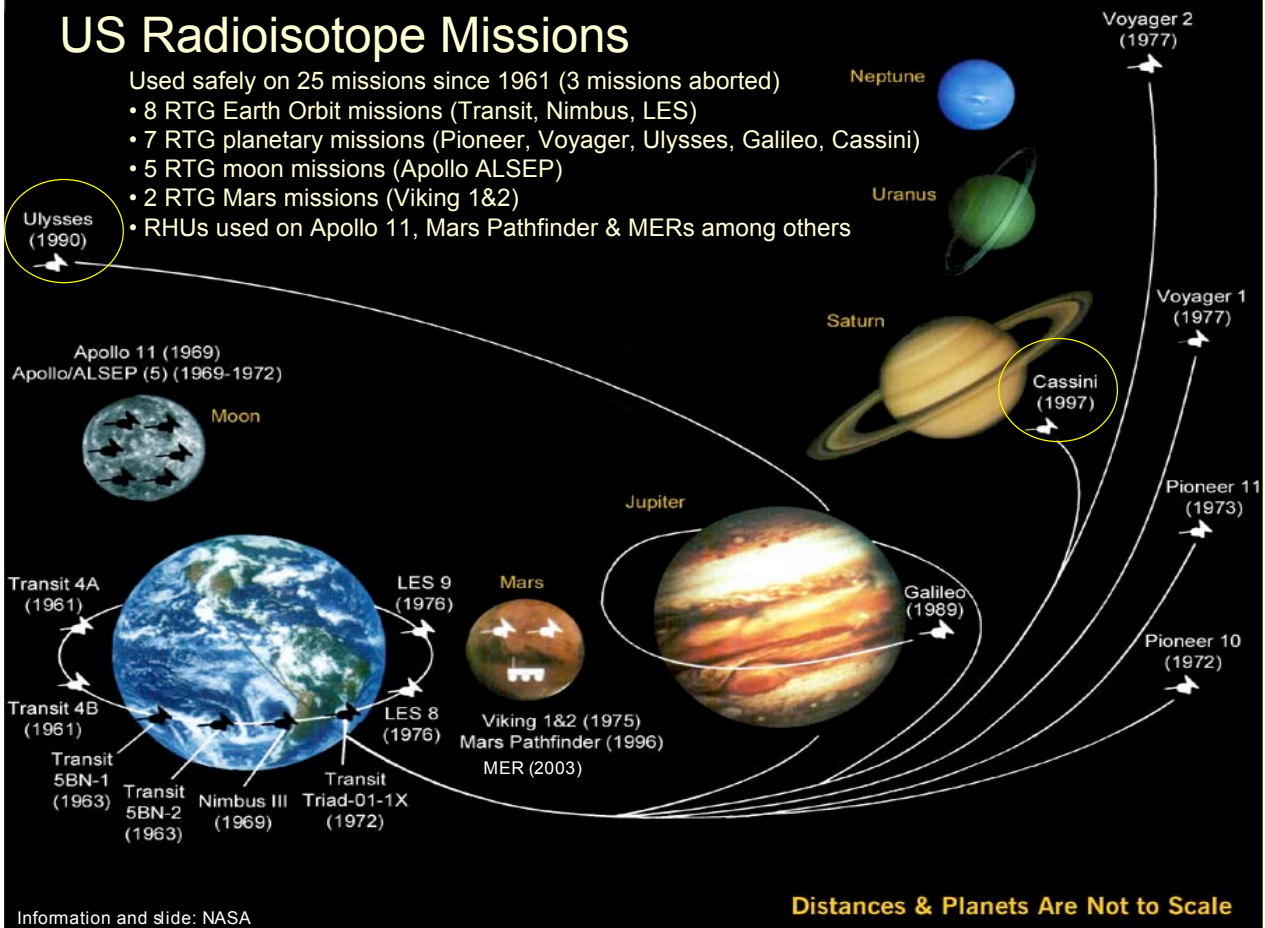
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# US Radioisotope Missions

Used safely on 25 missions since 1961 (3 missions aborted)

- 8 RTG Earth Orbit missions (Transit, Nimbus, LES)
- 7 RTG planetary missions (Pioneer, Voyager, Ulysses, Galileo, Cassini)
- 5 RTG moon missions (Apollo ALSEP)
- 2 RTG Mars missions (Viking 1&2)
- RHUs used on Apollo 11, Mars Pathfinder & MERs among others



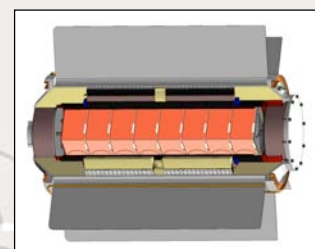
## 2.3 International Situation: US

### ③ Project Prometheus (bis):

#### ③ RTGs:

##### ③ Multi Mission RTG (MMRTG)

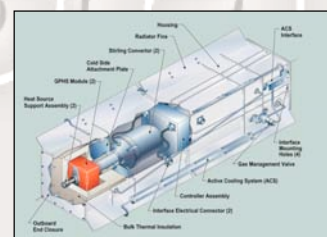
contract awarded to *Boeing (Boeing Rocketdyne + Teledyne Energy Systems)*;  
14 years lifetime, 3 years on Mars, space and planetary use;  
Martian rover, 110 We; 24-34 kg; TE system, ca 4 kg Pu-238,



MMRTG

##### ③ Stirling Radioisotope Generator (SRG)

contract awarded to *Lockheed Martin (+ Stirling Technologies and GRC)*,  
114/93 We, 27kg, fourfold reduction of Pu-238 need (1.2kg)



SRG

#### ③ Power conversion technology:

Competitively selected 10 contracts (50 M\$ total, 3 years) awarded



## 2.4 International Situation: US

### ③ Project Prometheus (bis):

- ③ Reactor research – JIMO:
  - ③ Reactor concept screening activity finished by DoE
  - ③ Power conversion by NASA: 3 competitive contracts awarded
  - ③ 3 competitive 10-month contracts awarded to industry for study of JIMO mission proposals (Boeing, Lockheed Martin, Northrop Grumman): (50 M\$ each)
  - ③ Naval Reactors responsible for nuclear core (currently different responsibility for conversion system!)



Artist impression of JIMO spacecraft (NASA)



## 2.5 International Situation: US

### Plans

- ③ 2.7 B\$ to be spent in 2004 – 2007 (680 M\$ annual average)
- ③ **Reactor development - JIMO:**
  - ③ 3 competitive 50 M\$ studies to be finished on 2004
  - ③ non-nuclear testing at MSFC (Early Flight Fission Test Facility); two main systems studied:
    - ③ Direct Drive Gas Cooled Reactor (DDG), Sandia design, fast spectrum U-235, noble gas
    - ③ Safe Affordable Fission Engine (SAFE), LANL design, fast spectrum U-235, heat pipes
- ③ **RTGs: (estimated down selection for SRG and MMRTG: 12/06)**
  - ③ SRG: qualification unit: 12/05  
Flight units: #1: 11/07, #2: 04/08, #3: 09/08
  - ③ MMRTG: qualification unit: 06/06  
Flight units: #1: 02/08, #2: 08/08
- ③ **Pu-238 production line:**
  - ③ “Record of Decision” signed in January 2001 for the “reestablishment of a domestic production capacity to produce 2 to 5 kg [of Pu-238] per year”
  - ③ Production line installation takes 5-6 years, total cost ~76 M\$ (extracted from presidential budget requests), final decision not taken as by July 2003





## 2.6 International Situation: US

### ⌚ Attempt of an analysis

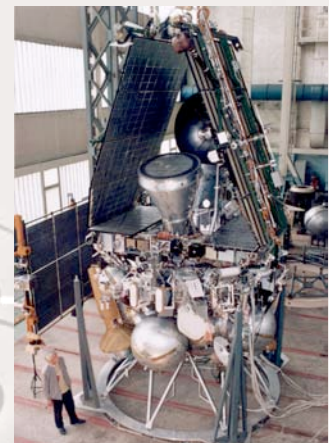
- ⌚ Technical and institutional capacity ✓
- ⌚ Pu-238 national production facility re-establishment start due by 2004/05 for national security purposes
- ⌚ Scientific exploration only justification for (high) development effort?
  - ⌚ Presence of strong national security interests/applications
- ⌚ Signs for sustained efforts
  - ⌚ Navy personnel involvement, clear objective (JIMO) and timeline
  - ⌚ NASA commitment *before* any DoE commitment
  - ⌚ No nuclear fuel development – realistic budgeting
- ⌚ Parallel industry contracts increase cost



## 2.7 International Situation: RU

### ⌚ Current Status – Russian situation

- ⌚ Early decision in favour of fission reactors
  - 35 reactors launched (mostly ocean reconnaissance missions);
- ⌚ Few RTGs
  - based on Po-210 for *Lunakhod* missions;
  - Small Pu-238 RTGs for *Mars96* mission;
- ⌚ Large nuclear thermal propulsion programme – no launch
  - test sites (partially?) intact
- ⌚ Ongoing Pu-238 production facility
- ⌚ Programmes stopped since 1 decade
  - Aging workforce – institutional competence?
- ⌚ TOPAZ II models sold to US in 1992 for potential cooperation



Mars96 (1996)



Topaz-1 spacecraft (1987)



## 2.8 International Situation: RU

### ☉ Currently most relevant available components:

- ☉ TOPAZ fission reactor (including associated subsystem technology, e.g. high temperature materials, large crystal growth technology; and 20 years development experience);
- ☉ Mini-RTGs and the Pu-238 production capacity;
- ☉ Nuclear test facilities.

### ☉ Flight-proven systems

- ☉ 33 Bouk/Romashka reactors (RORSAT ocean radar surveillance satellites).
  - ☉ 100s-We; fast neutron core, ; 20-25kg U-235 (90%); NaK cooling; thermoelectric conv.; reactor mass 390kg (excl. cold source); last launch 1988.
- ☉ 2 TOPAZ-1 (1987: Cosmos 1818; 1988: Cosmos 1867)
  - ☉ 5 kWe, epithermal core, 11.5 kg U-235 (96%); NaK cooling; thermionic in-core conv.; 0.5/1 year operations; reactor mass: 980 kg
  - ☉ TOPAZ-2: slight improvements (176 kg/kWe), never launched

### ☉ ISTC Studies (Int. Science and Technology Centre)

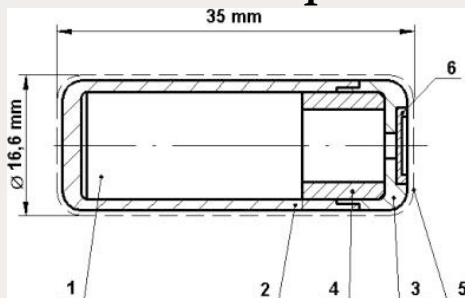
Two studies most valuable:

- ☉ ISTC-335: feasibility of a NTP development for a lunar tug application and potentially human Mars missions (1996/1997, CEA involvement);
- ☉ ISTC-1172: human Mars mission technology; 3 different nuclear power station designs for Martian surface reactors (100-125 kWe).



## 2.9 International Situation: Ru

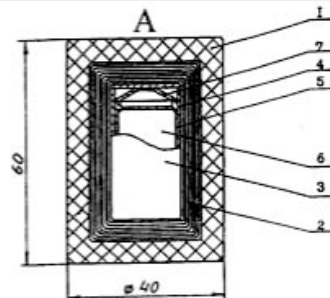
### ☉ Current and past Russian NPS devices: RHU/RTGs



- 1) 238Pu clad by iridium and packed into a casing made of a platinum-based alloy
- 2) protective shell case (tantalum-based alloy)
- 3) protective shell cover (tantalum-based alloy)
- 4) plug (tantalum-based alloy)
- 5) three-layer protective coating
- 6) membrane (tantalum-based alloy)

#### RHU (*Pantera*):

thermal power @BoL: ~8.5 W  
Mass: ~0.085 kg  
Dimensions: 16.7x35 mm cylinder



1. Outer protective canister
2. Composite protective canister
3. Radioactive heat source (RHS)
4. RHS protective shell
5. RHS capsule
6. Pu<sub>2</sub>O pellet (15-17 grams)
7. Spring

Power output: 8.5 watts  
Mass: 180 grams  
Height: 60 mm  
Diameter: 40 mm

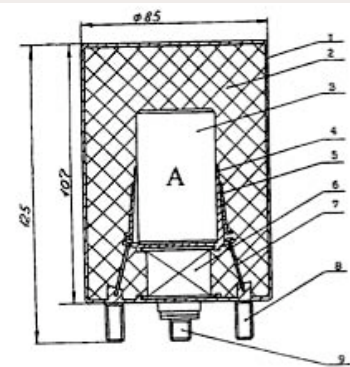




## 2.9 International Situation: Ru

### ④ Current and past Russian NPS devices: RTG

|  | spec. TY 95 11-82                       |   | spec. TT45.T.238-93 |
|--|---|---|---------------------|
|  | Mark A                                  | Mark B                                  |                     |
| $^{238}\text{Pu}$  |   |   |                     |
| mass fraction (not less)   | 67                                      | 59                                      | 74                  |
| isotope mass fractions:  |   |   |                     |
| $^{238}\text{Pu}$  | 80                                      | 70                                      | 85                  |
| $^{239}\text{Pu}$  |   |   |                     |
| $^{240}\text{Pu}$  |   |   |                     |
| $^{241}\text{Pu}$  | 20                                      | 30                                      | 15                  |
| $^{242}\text{Pu}$  |   |   |                     |
| $^{236}\text{Pu}$  |   |   |                     |
| mass fraction (not less)   | $2 \times 10^{-4}$ of $^{238}\text{Pu}$ | $3 \times 10^{-5}$ of $^{238}\text{Pu}$ | ?                   |
| mass fraction of $\text{Pu}$ isotope oxides (not less)                                     | 95                                      | 95                                      | -                   |
| mass fraction of actinide impurities in % of $\text{Pu}$ total mass (not more)             | -                                       | -                                       | 1                   |
| neutron flux of a preparation containing 1g $^{238}\text{Pu}$ , $\text{c}^{-1}$ (not more) | $2.5 \times 10^4$                       | $1.9 \times 10^4$                       | $2.5 \times 10^4$   |



- |                    |                             |
|--------------------|-----------------------------|
| 1. Metal Housing   | 6. Thermoelectric Converter |
| 2. Heat Insulation | 7. Excess heat sink         |
| 3. RTH             | 8. Mounting Bolts           |
| 4. Bracket         | 9. Power Socket             |
| 5. RTH Holder      |                             |

Power output: 0.25 watts  
 Mass: 0.555 kg  
 Height: 125 mm  
 Diameter: 85 mm



## 2.10 International Situation: RU

### ④ Attempt of an analysis

- ④ Rich heritage from Soviet developments
- ④ Only available and flight proven reactor system
  - ④ Reason for complete halt of national use unclear (cost only?, competence/facilities no longer within Russia?)
- ④ Aging workforce due to stopped national programmes – decreasing institutional competence?
- ④ Open for cooperation
  - ④ willingness to share core technical and technological competence (e.g. on materials) questionable; lessons from TOPAZ-2 cooperation in 1990s?



## 2.11 International Situation: *others*

- ③ **No information available on Japanese, Indian, Brazilian NPS activities;**
  - ④ NPS activities within UN COPUOS are closely followed by Argentina, Pakistan, China (in addition to traditional space/nuclear powers)
  - ④ Reported Japanese study on NPS in late 1980s (unconfirmed)
- ③ **Likelihood of existing plans in China**
  - ④ Necessary for ambitious lunar missions
  - ④ Presence of chinese naval submarine reactors indicate adequate technological competence



## 3.1 European Situation

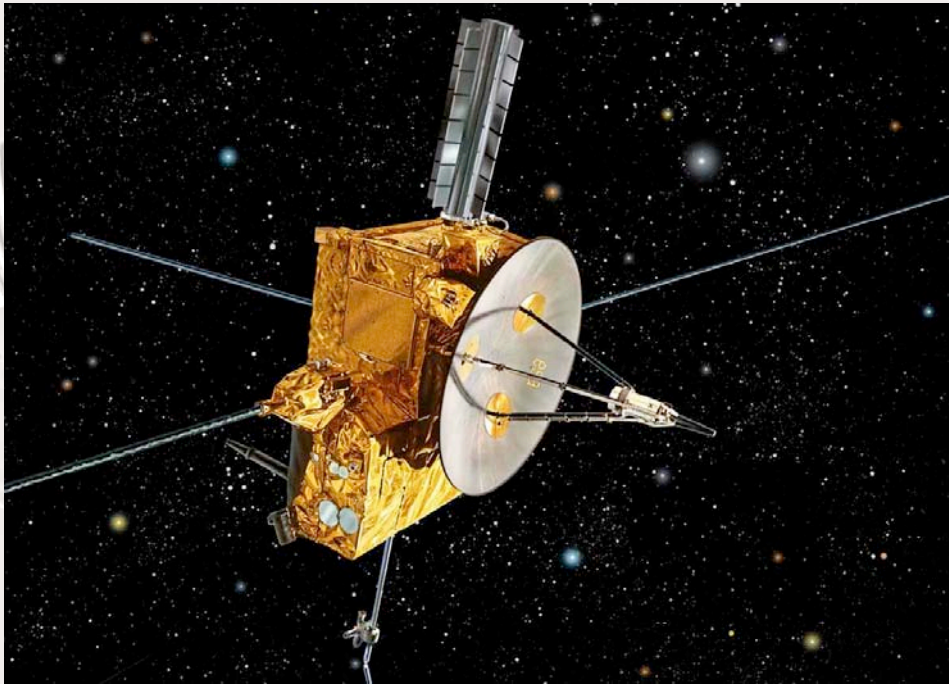
- ③ **No industrial experience with radioisotope thermal generators or in space nuclear fission reactors;**
- ③ **Closest industrial expertise in field of naval submarine nuclear reactors (France);**
- ③ **Major reactor concept studies in 1980/90s: ERATO, MAPS, UKSR;**
- ③ **Adhesion of East-European countries might change situation**
  - ④ TOPAZ assembly plant near Tallinn, Estonia, competence remaining?
- ③ **Use of US NPS (RTGs and RHUs) in common scientific exploration missions**
- ③ **Recent studies on space NPS within GSP**



Cassini/Huygens spacecraft



## 3.2 Ulysses Spacecraft



## 3.3 European Situation - France

- ① **1983-89: ERATO programme start; feasibility of a 20-400 kWe system, 1995-97: MAPS nuclear thermal propulsion system, 300 MWth system;**
- ① **competence maintained (on small scale) within the CEA**
  - ① CNES séminaire on NPS in Sept. 2002 (~100 participants, no concrete projects).
  - ① There seems to be no national civil or defence project on space NPS at the moment.
- ① **Industrial competence from**
  - ① defence projects: naval reactors, CEA-Technicatome;
  - ① large civil nuclear power complex: research, logistics, maintenance, handling, security, e.g. COGEMA, COGEMA Logistics, ... ;
  - ① propulsion system (SNECMA).
- ① **Inspection and study of TOPAZ reactor system in 1990s – abandoned, lessons learnt?**
- ① **Probable assessment of establishing a national RTG production in 1980/90s with negative outcome (unconfirmed)**





## 3.4 European Situation – UK,D,I

### ③ UK:

- ③ 1987 start of UK Space Reactor (UKSR) studies, He/Xe cooled high temperature reactor; relying on “Dragon” project experience (fuel, high temperature).

### ③ Germany:

- ③ late 1960s/beginning 1970s: first European space reactor study “*In Core Thermionic Reactor*”; ended with US SNAP programme end;
- ③ academic competence available (associated industrial capacity probably partly available);
- ③ national nuclear exit strategy: ⇒ significant involvement of Germany in any space nuclear power sources development seems rather unlikely;
- ③ Option: a European (civil) space nuclear programme could provide an opportunity to continue national nuclear developments and preserve industrial nuclear research capacity.

### ③ Italy:

- ③ Renewed interest in space NPS via Rubbia Am-242 concept; ASI involvement, significant national spending;
- ③ Principal decision against terrestrial nuclear power stations still valid; academic research on small scale.



## 3.5 European Situation – defence

### ③ Defence sector

- ③ 1980s: studies in France and GB at least partly defence motivated (F: orbital space plane);  
no information on current projects or studies.



## 3.6 European Situation – ESA

### Industrial Activities – General Studies Programme

#### ⌚ Reactor studies:

- ⌚ Technicatome “Lunar Nuclear Power System (LuNPS)”, 12441/97/NL/MV
  - ⌚ Feasibility of a lunar surface reactor based on existing European reactor technology; highly enriched PWR; thermoelectric conversion and radiators; 10 kWe.
- ⌚ QinetiQ (subcontractors: Technicatome, AEA Techn., Serco Ass.) “Future Power Systems for Space Exploration”, 14565/00/NL/WK; 2000-2003
  - ⌚ Assessment of power systems for Mars exploration; Surface power reactors; NEP and NTP;
  - ⌚ 2 separate reactors for propulsion and surface power
  - ⌚ Development of two 50 kWe Martian surface reactor systems:
    - ⌚ liquid metal – thermoelectric – radiator Technicatome design
    - ⌚ gas-cooled – Brayton – CO<sub>2</sub>-convection QinetiQ design
- ⌚ Politecnico di Milano: “Space nUclear REactor - SURE” study, 17301/03/NL/LvH, 2003
  - ⌚ synergies between terrestrial and space reactors, minimum development cost approach, assessment of relevant Italian nuclear capacities
- ⌚ ESA “Propulsion 2000” programme including nuclear propulsion assessments
- ⌚ SEP-NEP study
- ⌚ “Photonic Propulsion for Interplanetary and Outer Solar System Missions” TOS-MPE study in preparation



## 3.7 European Situation – ESA

#### ⌚ RTG studies:

- ⌚ “Radioisotope Energy Storage Assessment”, study by AEA Technology in 1996
- ⌚ “Radioisotope Thermal Generators for LEDA Programme”, study by Krasnaya Zvezda (Ru) in 1996
  - ⌚ study on the availability of Russian RTGs for the ESA lunar programme
- ⌚ “BLAPOS RTG-200”, study currently under negotiation with BLAPOS (Ru)
  - ⌚ study on the possibility to upgrade existing Russian small RTGs to power levels between 50 and 200 We.

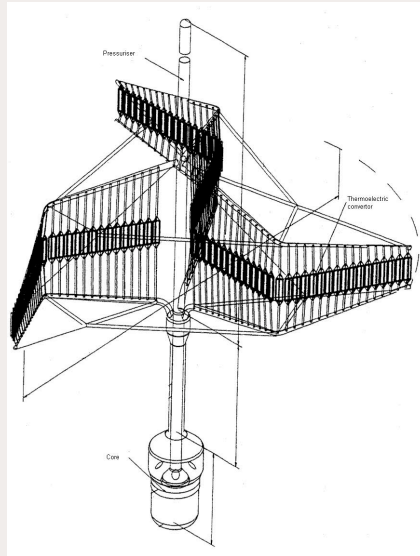
#### ⌚ Conversion system studies:

- ⌚ “MHD Converters” study by Alcatel ETCA (B), Technicatome (F), LEGI CNRS (F), CSL (B) in 2002/3
  - ⌚ metal gas and thermo-acoustic converters, 230kEuro
- ⌚ “Alternative Electrical Energy Generation with Thermoelectric Systems”; study by *Ecole des Mines*, Nancy

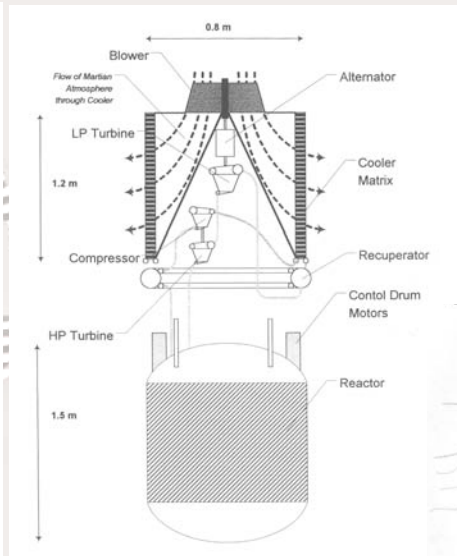


## 3.8 European Situation – *ESA/Industry*

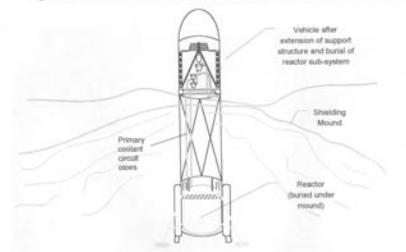
### Industry Reactor Designs



Liquid Metal – TE – Radiator concept  
(*Technicatome*)



Gas cooled – Brayton – forced convection concept  
(*QinetiQ*)



images QinetiQ



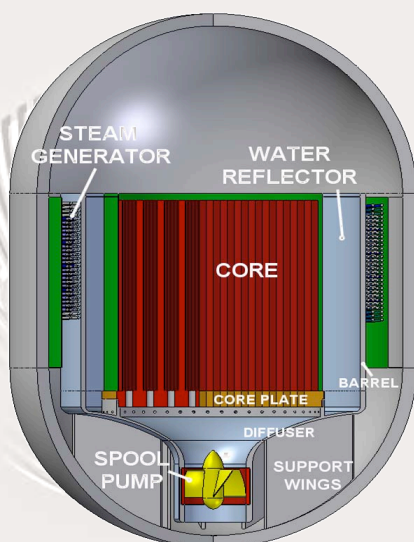
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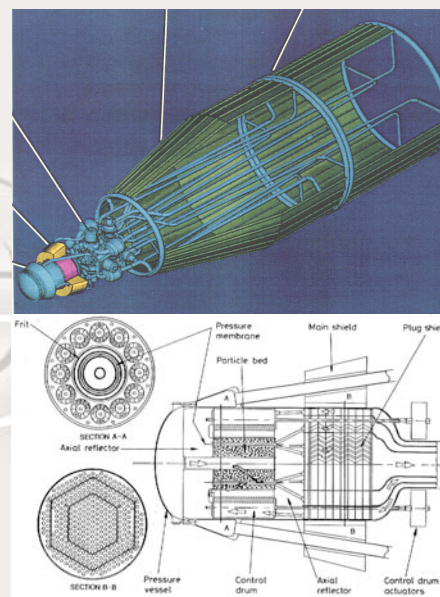
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## 3.9 European Situation – *ESA/Industry*



Space modified PWR – radiator  
(*SURE – PoliMi design*)



Gas-cooled – particle bed – radiator  
(*SNPS-200; QinetiQ*)



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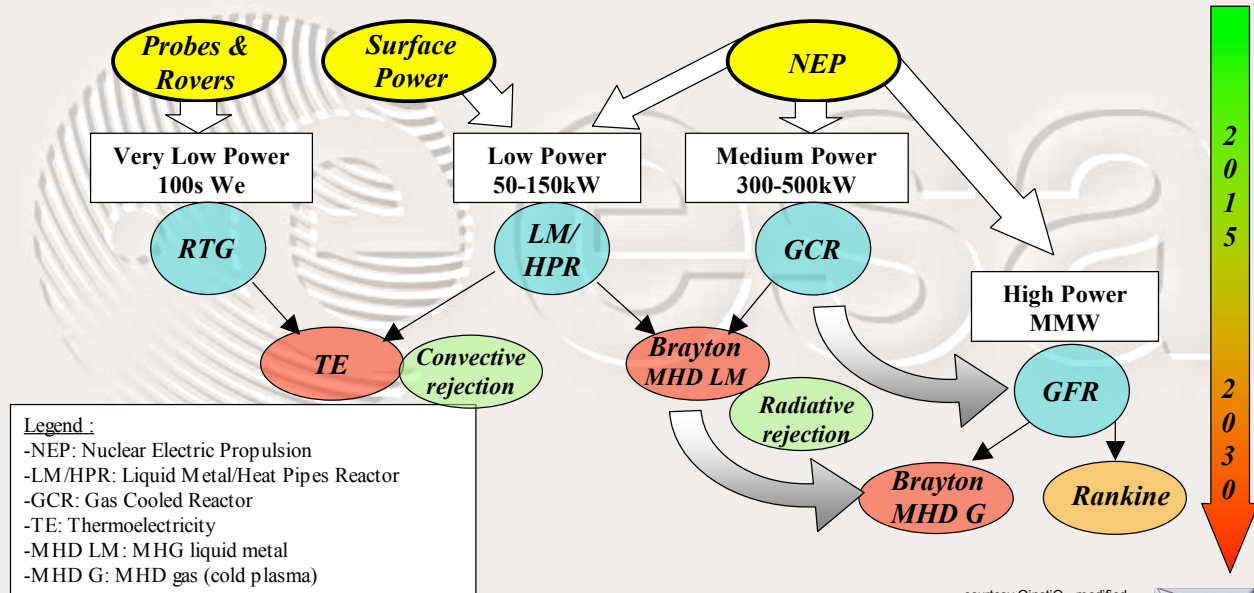
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## 3.10 European Situation – ESA/Industry

### Industry proposed general roadmap logic



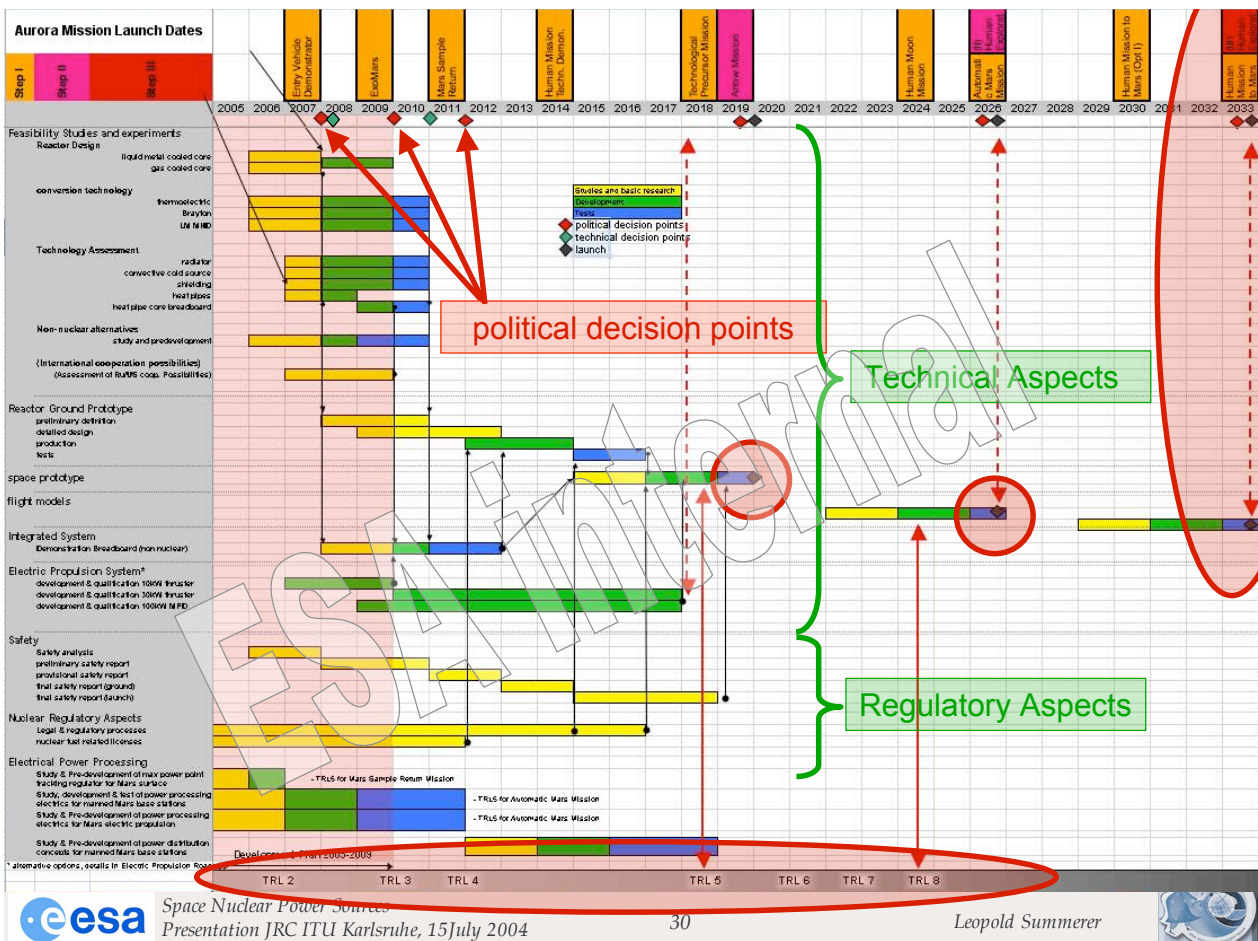
courtesy QinetiQ - modified



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## 3.12 European Situation – *ESA*

### ESA Internal Activities

- ③ **Science:**
  - ③ assessment of RTGs for different science missions
  - ③ TRP proposal “NEP systems for outer planet missions”
- ③ **SER (ACT):**
  - ③ assessment of legal aspects of the use of NPS
  - ③ assessment of administrative and logistic aspects of the use of (Russian) RTGs in Europe (*COGEMA Logistics*)
  - ③ integration of surface reactor aspects into human Mars mission CDF study
  - ③ participation in COPUOS NPS working group, participation in establishment of multi-year work plan
- ③ **Recent ESA NPS meetings**
  - ③ Jan 02: Industry/CEA/ESA meeting at ESTEC
  - ③ Jun 03: CNES/ASI/ESA meeting at ESA HQ
  - ③ Jan 04: ESA HQ meeting on NPS strategies
- ③ **EC/ESA NPS WG**
  - ③ 6 May 04: 1st meeting (establishment)
  - ③ 30 June 04: 2nd meeting (agreement on mandate and workplan)



## 3.13 European Situation

- ③ **Attempt of an analysis**
  - ③ Principal availability of industrial and technical capacity ✓
  - ③ Availability of independent NPS launch capability ✓
  - ③ Long term: need of NPS for space exploration, possible “usefulness” for military space applications
  - ③ independent development of RTGs unlikely
  - ③ research on terrestrial nuclear reactors ongoing in Europe while halted in US for several years ⇒ respected partner
  - ③ European decision structure possibly more favourable to long-term NPS effort than annual congressional approval system in US
  - ③ Work on space reactors presents strong “academic attraction” (lack of young nuclear scientists)
  - ③ Absence of an appropriate institutional frame / Absence of a European DoE/DoD/DoT
  - ③ Assessment of “non-space” interests in NPS developments (education, high temperature materials, Gen IV activities ...)



## 3.14 Possible European NPS Needs

### European Exploration Programme:

- ⌚ Robotic Mars Missions
  - ⌚ e.g. ExoMars 2009:
    - RHUs/RTGs for Mars rover under serious consideration (e.g. Russian supply)
    - Phase B trade-off as of 2004/05
- ⌚ Robotic / Human Moon missions
  - ⌚ Survival of 14 Earth-days long lunar night (2020 timeframe)
- ⌚ All missions beyond Mars
  - ⌚ Rosetta mission: demonstration of limits of “non-nuclear” spacecraft
- ⌚ Human Mars Missions (2030+ timeframe)
  - ⌚ Nuclear reactors for surface stay / Mars base missions - power supply
  - ⌚ Nuclear reactors to shorten transfer time - nuclear (electric) propulsion



## 4.1 Potential Role of EC JRC ITU

### ⌚ Relevant open questions:

- ⌚ In case of use of Russian radioisotope sources:
  - ⌚ Who owns them? Responsibility
  - ⌚ Who transports them?
  - ⌚ Who manipulates them? Where? What facilities are already available?
  - ⌚ Who tests them? Where? What facilities are already available? Who can verify Russian test data?
  - ⌚ Who licences them?
  - ⌚ Who establishes the approval/licensing procedures?  
(Who plays the European DoE?)
- ⌚ Interest for these developments outside from space?





## 4.2 US example - repartition of tasks

